# A Novel Approach for Predicting Frictional Factor during Fluid Flow In well Tubings and Flowlines.

#### Akinade Akinwumi, Omohimoria Charles

Abstract - One of the main challenges facing by production engineers is the ability to choose an accurate friction factor during a fluid flow, it was observed that most of the existing models (Fanning, 1944), Blasius, 1914) and moody e.t.c) over-predict the values of frictional factors when compare with the experimental value, this is due to the fact that these models are dependent on Reynolds number and pipe roughness, and since pipe roughness changes with use, this make them inaccurate, it is therefore important to develop a new correlation that will be a function of Reynolds number and that can accurately matched with experimental data in order to aid accurate prediction of wax deposition and thickness during crude oil production. In this research work an accurate frictional factor model (correlation) was developed by considering the universal velocity profile (distribution) over the cross section of a pipe during the lamina and turbulent flow, the model result was tested and validated with complex cases.

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Keywords: Friction factor, Reynold number, Velocity profile, Laminar flow

## 1. INTRODUCTION

The basis for the application of Wax thickness model starts from the development of Friction factor as a function of Reynolds number, due to the result published by Wax Joint Industry Project (Wax JIP) sponsored by US Department of Energy, the research work was carried out at the University of Tulsa. It was published that wax interface can be assumed smooth, since a significant amount of oil is usually trapped in the wax layer. This smooth wax interface was confirmed by observation of the wax layer in a spool piece removed from the Wax JIP single phase flowline during a preliminary test. Therefore the friction factor models can be developed by considering the velocity profile distribution in a laminar and turbulent flow pipe, from which friction factor as a function of Reynolds number will be developed for both laminar and turbulent flow.

In general there are three types of fluid flow in pipes; laminar, turbulent and transitional flow. Laminar flow generally happens when dealing with small pipes, low flow velocities and with highly viscous fluids. At low velocities fluids tend to flow without lateral mixing, and adjacent layers slide past one another like playing cards. There are neither cross currents nor eddies. Laminar flow can be regarded as a series of liquid cylinders in the pipe, where the innermost parts flow the fastest, and the cylinder touching the pipe isn't

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moving at all. In turbulent flow, the fluid moves erratically in the form of cross currents and eddies. Turbulent flow happens in general at high flow rates and with larger pipes. Transitional flow is a mixture of laminar and turbulent flow, with turbulence in the center of the pipe, and laminar flow near the edges.

It has long been known that in turbulent flow a rough pipe leads to a larger friction factor for a given Reynolds number than a smooth pipe does. If a rough pipe is smoothed, the friction factor is reduced. When further smoothing brings about no further reduction in friction factor for a given Reynolds number, the tube is said to be hydraulically smooth.

In turbulent flow, the friction factor, f depends upon the Reynolds number and on the relative roughness of the pipe, k/D, where, k is the roughness parameter(average roughness height of the pipe) and D is the inner diameter of the pipe. The general behavior of turbulent pipe flow in the presence of surface roughness is well established. When k is very small compared to the pipe diameter D i.e.  $k/D \rightarrow 0$ , f depends only on N<sub>Re</sub>. When k/D is of a significant value, at low N<sub>Re</sub>, the flow can be considered as in smooth regime (there is no effect of roughness). As N<sub>Re</sub> increases, the flow becomes transitionally rough, called as transition regime in which the friction factor rises above the smooth value and is a function of both k and N<sub>Re</sub> and as N<sub>Re</sub> increases more and more, the flow eventually reaches a fully rough regime in which f is independent of N<sub>Re</sub>.

In a smooth pipe flow, the viscous sub layer completely submerges the effect of k on the flow. In this case, the friction factor f is a function of  $N_{Re}$  and is independent of the effect of k on the flow. In case of rough pipe flow, the viscous sub layer thickness is very small when compared to roughness height and thus the flow is dominated by the roughness of the pipe

wall and f is the function only of k/D and is independent of  $N_{\text{Re}}$ 

It has been observed that most of the deposition models that were developed by earlier researchers over predict the location and rate of wax build up in flowlines when compared with experimental results, this is as a results of their derivation from a faulty thermodynamic principles (or models), it is therefore imperative to develop a viable and reliable deposition model that can accurately predict wax deposition rate in flowlines when compared with experimental data by using an accurate friction factor during modeling of such equation. One of the main challenges facing by production engineers is the ability to choose an accurate friction factor during a fluid flow, it was observed that most of the existing models (Fanning(1944), Blasius (1914) and moody e.t.c) overpredict the values of frictional factors when compare with the experimental value, this is due to the fact that these models are dependent on Reynolds number and pipe roughness, and since pipe roughness changes with use, this makes them inaccurate, it is therefore important to develop a new correlation that will be a function of Reynolds number and that can accurately matched with experimental data in order to aid accurate prediction of wax deposition and thickness during crude oil production.

In this research work an accurate frictional factor model (correlation) was developed by considering the universal velocity profile (distribution) over the cross section of a pipe during the lamina and turbulent flow, the model results was validated by subjecting it to a complex flow situations.

# 2.0 MODEL FORMULATION

One of the main challenges facing by production engineers is the ability to choose an accurate friction factor during a fluid flow, it was observed that most of the existing models (Fanning(1944), Blasius (1914) and moody e.t.c) over-predict the values of frictional factors when compare with the experimental value, this is due to the fact that these models are dependent on Reynolds number and pipe roughness, and since pipe roughness changes with use, this makes them inaccurate, it is therefore important to develop a new correlation that will be a function of Reynolds number and that can accurately matched with experimental data in order to aid accurate prediction of wax thickness in flowline.

#### (A) MODEL DEVELOPMENT

This can be achieved by consideration of universal velocity profile (distribution) over the cross section of a pipe during the lamina and turbulent flow

## FOR LAMINAR FLOW

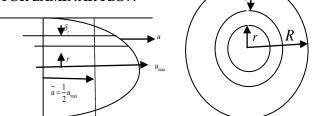


Fig 3.2: Velocity distribution in laminar flow pipe

To calculate the volume flow rate through a pipe of diameter d in term of a pressure drop over a length L we apply:

# **Poiseuille Equation**

$$Q = \frac{\Delta P \pi d^4}{128 \mu L} \tag{1}$$

but head loss due to friction is given as

$$h_f = \frac{\Delta P}{\rho g} \tag{2}$$

$$=\frac{128\mu LQ}{\rho g} \tag{3}$$

now:

$$\mathbf{Q} = \pi \left(\frac{d^4}{4}\right) u \tag{4}$$

applying Darcy equation for head loss in circular pipe

$$h_{f} = \frac{128\mu L\pi d^{2}u}{4\rho g\pi d^{4}} = \frac{4\rho Lu^{2}}{2gd}$$
(5)

$$\varphi = \left(\frac{128\mu}{16\rho ud}\right) \tag{6}$$

recall that

or

$$R_{e} = \frac{\rho \mu d}{\mu} \tag{7}$$

Substitute into equation 6

$$\varphi = \frac{8}{R_e}$$

$$\varphi = 8R_e^{-1} \tag{8}$$

The above equation is plotted as a straight line on log-log plot and is independent of surface roughness.

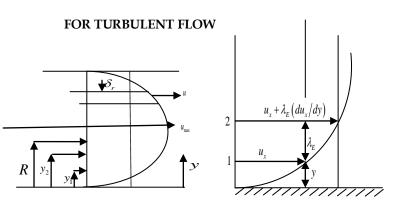


Fig 3.3a: Velocity distribution in turbulent flow pipe Fig 3.3b: Prandtl mixing length

Fig 3.3b above shows the velocity profile near a surface.

At point 1, the velocity is  $u_x$  and at point 2 the velocity is  $u_x^1$ . For an eddy velocity  $u_{Ey}$  in the direction perpendicular to the surface, the fluid is transported away from the surface at a mass rate per unit area equal to  $u_{Ey}\rho$ , this fluid must be replaced by an equal mass of fluid which is transferred in the opposite direction.

The momentum transferred away from the surface per unit time is given as

$$R_{y} = \rho u_{Ey} \left( u_{x} - u_{x}^{1} \right)$$
(9)

If the distance between the two location is approximately equal to the mixing length  $\lambda_E$ , and if the velocity gradient is nearly constant over the distance

$$\frac{u_x - u_x}{\lambda_E} \approx \frac{du_x}{dy} \tag{10}$$

Assuming again that  $u_{x}^{T} - u_{x} \approx u_{Ey}$ 

$$R = \rho \lambda_E^2 \left(\frac{du_x}{dy}\right)^2 \tag{11}$$

it is assumed throughout that no mixing take place with the intervening fluid when eddy transport fluid element over a distance equal to the mixing length close to the surface  $R_v \rightarrow R_0$ , and  $\overline{R}$ 

$$\frac{\mathbf{R}}{\rho} = \lambda_E \frac{du_x}{dy} \tag{12}$$

 $\sqrt{\frac{\kappa}{\rho}}$  is known as shearing stress velocity of

frictional velocity and it is usually denoted by  $u^*$ In steady state flow over a plane surface, or close to the wall for flow in a pipe,  $u^*$  is constant and equation 3.37 can be integrated provided that the relation between  $\lambda_E$  and y is known.  $\lambda_E$ will increase with y and, if a linear relationship is assumed then

$$\lambda_E = k_{y} \tag{13}$$

$$u^* = k_y \frac{du_x}{dy} \tag{14}$$

On integration

$$\frac{u_x}{u^*} = \left(\frac{1}{k}\right) \ln\left(\frac{yu^*\rho}{\mu}\right) + B$$
(15)

Where B is a constant.

Or 
$$\frac{u_x}{u^*} = \left(\frac{1}{k}\right) \ln\left(\frac{yu^*\rho}{\mu}\right) + B^1$$
 (16)

Since  $\frac{u^* \rho}{\mu}$  is a constant, B<sup>1</sup> will also be constant.

Writing the dimensionless velocity term  $\frac{u_x}{u^*} = u^+$ and the dimensionless derivatives as  $\left(\frac{yu^*\rho}{\mu}\right)y = y^+$  $u^+ = \frac{1}{k}\ln y^+ + B^1$  (17)

if equation 7 is applied to the outer edge of the boundary layer, when  $y = \delta$  (boundary layer thickness) and  $u_x = u_s$  (stream velocity) then:

$$\frac{u_s}{u^*} = \frac{1}{k} \ln \frac{\delta u^* \rho}{\mu} + B^1$$
(18)

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Subtracting equation (16) from (18)

$$\frac{u_s - u_x}{u^*} = \frac{1}{k} \ln \frac{\delta}{y}$$
(19)

Using experimental result results for flow of fluid over both smooth and rough surfaces Nikurade found K to have a value of 0.4

Thus:

$$\frac{u_s - u_x}{u^*} = 2.5 \ln \frac{\delta}{y} \tag{20}$$

For fully developed flow in pipe  $\delta = r$  and  $u_s$  is the velocity at the axis and then

$$\frac{u_s - u_x}{u^*} = 2.5 \ln \frac{r}{y} \tag{21}$$

equation 4.36 is known as velocity defect law. Note that for turbulent core the value of 0.4 can be substituted for in equation 3.42 to give:

$$u^+ = 2.5 \ln y^+ + B^1$$

From the plot of  $u^+$  and  $y^+$ . It is observed that for a smooth surface  $B^1 = 5.5$ . Thus for a smooth pipe  $u^+ = 2.5 \ln y^+ + 5.5$  (23)

Substituting for  $u^+ = \frac{u_s}{u^*}$  and  $y^+ = \frac{r\rho u^*}{\mu}$  into equation 3.47

Thus: 
$$u_s = u^* \left( 2.5 \ln \frac{r \rho u^*}{\mu} + 5.5 \right)$$
 (24)

but it also necessary to obtain expression for mean velocity U of the fluid, we use the relation

$$u = \int_{0}^{r} \frac{\left(2\pi \left(r - y\right) dy u_{x}\right)}{\pi r^{2}}$$
(25)

$$=2\int_{0}^{1}u_{x}\left(1-\frac{y}{r}\right)d\left(\frac{y}{r}\right)$$
(26)

Substituting for  $u_x$  equation 3.50

$$u = 2\int_{0}^{1} \left( u_x + 2.5u^* \ln \frac{r}{y} \right) \left( 1 - \frac{r}{y} \right) d\left( \frac{y}{r} \right)$$
(27)

$$\frac{u}{u_s} = 2\int_0^1 \left( u_x + 2.5 \frac{u}{u_s} \ln \frac{y}{r} \right) \left( 1 - \frac{y}{r} \right) d\left( \frac{y}{r} \right)$$
(28)

$$=2\left\{\frac{y}{r}-\frac{1}{2}\left(\frac{y}{r}\right)^{2}\right\}_{0}^{1}+5.0\frac{u^{*}}{u_{s}}\left\{\left(\ln\frac{y}{r}\right)\left[\frac{y}{r}-\frac{1}{2}\left(\frac{y}{r}\right)^{2}\right]\right\}$$

$$-\int_{0}^{1} \left(\frac{y}{r}\right)^{-1} \left[\frac{y}{r} - \frac{1}{2} \left(\frac{y}{r}\right)^{2}\right] d\left(\frac{y}{r}\right)$$
(30)

$$=1+5.0\frac{u^{*}}{u_{s}}\left\{0-\left(\frac{y}{r}\right)-\frac{1}{4}\left(\frac{y}{r}\right)^{2}\right\}_{0}^{1}$$
(31)

$$=1+5\frac{u}{u_{s}}\left(-\frac{3}{4}\right)=1-3.75\left(\frac{u}{u_{s}}\right)$$
(32)

Substituting into equation (24)

$$u = 3.75u^* = u^* \left\{ 2.5 \ln\left[\left(\frac{d\rho u}{\mu}\right)\left(\frac{r}{d}\right)\left(\frac{u^*}{u}\right)\right] + 5.5 \right\}$$
(33)

Since 
$$R_e = \frac{d\rho u}{\mu} = \frac{u_{\text{max}}}{u}$$
 when  $r = d$   
 $u + 3.75u^* = u^* (5.5)$   
 $u + 3.75u^* = 5.5u^*.$  (34)

Dividing through by  $u^*$ 

$$\frac{u}{u^*} = 5.5 - 3.75 \tag{35}$$

Since 
$$\frac{u}{u^*} = R_e \varphi^4$$

Therefore

$$R_e \varphi^4 = 1.75$$
 (36)

$$\varphi^4 = \left(\frac{1.75}{R_e}\right)^{\frac{1}{4}}$$

Therefore  $\varphi = 0.0396 R_e^{-\frac{1}{4}}$ .

Finally,

 $\varphi = 0.0396 R_e^{-0.25} \tag{37}$ 

Comparing equation (8) and (37), that is

 $\varphi = 8R_e^{-1}$  For laminar flow

 $\varphi = 0.0396 R_e^{-\frac{1}{4}}$  For turbulent flow

We can come out with a combined model for our predicted friction factor

$$\varphi = x R_e^{-y} \tag{38}$$

Where

x = 8 and y = 1 for laminar flow

x = 0.0396 and y = 0.25 for turbulent flow

# 3.0 MODEL VALIDATION

The model results is then compared with an experimental values obtained from the work of H.S. Fogler et al. (2000) the

results are tabulated in Table 4.4 and shown pictorially in Figure 4.3

TABLE 4.4: Comparison between Friction factor for Model output and experimental data

			MODEI
	REYNOLDS NO	EXPERIMENTAL DATA	MODEL RESULTS
	2500	0.00522	0.0056
	3000	0.00525	0.00535
	4000	0.00475	0.00498
	5000	0.0045	0.00471
	6000	0.00425	0.0045
	7000	0.004	0.00433
	8000	0.003875	0.00419
	9000	0.00375	0.00407
	10000	0.003625	0.00396
	20000	0.00325	0.00333
	30000	0.00275	0.00301
	40000	0.002625	0.0028
	50000	0.0025	0.00265
	60000	0.002375	0.00253
	70000	0.00225	0.00243
	80000	0.0021875	0.00235
	90000	0.002125	0.00229
	100000	0.0020625	0.00223
	200000	0.00185	0.00187
	300000	0.00165	0.00169
	400000	0.00155	0.00157
r	500000	0.00145	0.00149
r	600000	0.0014	0.00142
	700000	0.00135	0.00137
	800000	0.00132	0.00132
	900000	0.00128	0.00129
	1000000	0.00123	0.00125

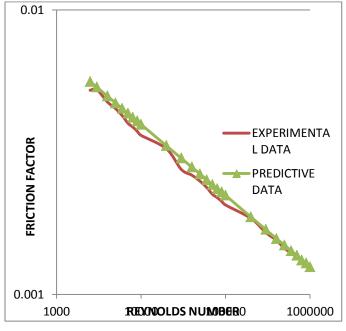


Fig 4.3: Comparison between Friction Factor of Experimental Data and Model Results

Moreover the Friction factor model output for turbulent flow (equation 3.60) is further validated by comparison with smooth pipe equation for turbulent flow of the other existing research work derived by Fanning, Blasius or Moody. The results are presented in the Table 4.4 and Figure 4.4 below.

TABLE 4.5: Comparison of Model output with other existing models (Fanning & Blasius)

models (Fanning & Blasius)					
RENOLD	EXPERIMENTA	MODEL	FARNNING	BLASIUS	
NO	L DATA	RESULT	FRICTION FACTOR	FRICTION	
				FACTOR	
2500	0.00522	0.0056	0.0112		
3000	0.00525	0.00535	0.0107	0.0427	
4000	0.00475	0.00498	0.00993	0.0397	
5000	0.0045	0.00471	0.0094	0.0376	
6000	0.00425	0.0045	0.00898	0.036	
7000	0.004	0.00433	0.00864	0.0345	
8000	0.003875	0.00419	0.00835	0.0334	
9000	0.00375	0.00407	0.00811	0.0324	
10000	0.003625	0.00396	0.0079	0.0316	
20000	0.00325	0.00333	0.00664	0.0266	
30000	0.00275	0.00301	0.006	0.024	
40000	0.002625	0.0028	0.00559	0.0223	
50000	0.0025	0.00265	0.00528	0.0211	
60000	0.002375	0.00253	0.00505	0.0202	
70000	0.00225	0.00243	0.00486	0.0194	
80000	0.0021875	0.00235	0.0047	0.0188	
90000	0.002125	0.00229	0.00456	0.0182	
100000	0.0020625	0.00223	0.00444	0.0178	
200000	0.00185	0.00187	0.00374	0.0149	
300000	0.00165	0.00169	0.00338	0.0135	
400000	0.00155	0.00157	0.00314	0.0126	
500000	0.00145	0.00149	0.00297	0.0119	

600000	0.0014	0.00142	0.00284	0.0114
700000	0.00135	0.00137	0.00273	0.0109
800000	0.00132	0.00132	0.00264	0.0106
900000	0.00128	0.00129	0.00256	0.0103
1x10 <sup>6</sup>	0.00123	0.00125	0.0025	0.00999

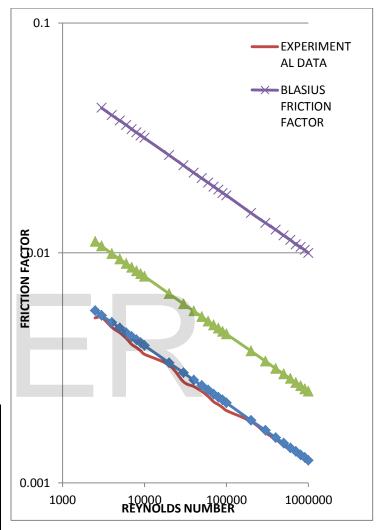


Fig 4.4: Comparison between Model output, Experimental data and other existing Models (Blasius and Fanning)

## **4.0 DISSCUSION OF RESULT**

The results of the model were compared with those obtained experimentally in Table 4.4; this is also shown graphically in Figure 4.3. It can be seen that the model output is in agreement with experimental result. The maximum absolute deviation was in the range of 0.00038, which in other word confirm the accuracy of the newly developed model.

The accuracy of the friction factor of the model output is further tested by comparison with friction factor for a smooth pipe of the other existing model as seen in Table 4.5, it can be seen in Figure 4.4 that there is a wide deviation between the other existing model and experimental value, where as there is

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close agreement between model output and experimental data, which confirm the accuracy of modeling a friction factor from velocity distribution profile in a flow pipe.

# **5.0 CONCLUSION**

The approach employed in this work is easily accessible since the application requires constant thermodynamic data (properties that varies with temp) and rheological properties of the crude.

The following conclusion can be deduced from this research work.

- 1. An accuracy modeling of a friction factor from velocity distribution profile in a flow pipe as been established
- 2. A friction factor model that is a function of surface roughness has been established.

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